

240(10) : Collected Results of Relativistic Conversions
to Some Formulae of Fitzpatrick

Converting the formula:

$$V(r) = -\frac{mMG}{r} \left(1 + \frac{E}{r^2}\right) \quad - (1)$$

for a symmetric top it is found that:

$$V(r) = -\frac{mMG}{r} \left(1 + \frac{E}{r^2}\right) - \frac{L_0^2 MG}{mc^2 r^3} \quad - (2)$$

where $L_0^2 \sim rm^2 MG \quad - (3)$

for a nearly circular orbit. Here:

$$E = R \Delta R \quad - (4)$$

where R is the equatorial radius and ΔR the difference between the equatorial and polar radii.

The perihelia precession from eq. (1) is:

$$\Delta\theta = 6\pi \left(\frac{E}{r^2} + \frac{MG}{c^2 r} \right) \quad - (5)$$

Denote $\Delta\theta(\text{slate}) = \frac{6\pi E}{r^2} \quad - (6)$

$$\Delta\theta(EGR) = \frac{6\pi MG}{c^2 r} \quad - (7)$$

Apply eqs. (6) and (7) to three systems:

2) 1) Earth / sun ; 2) earth / moon ; 3) earth / gravity
Probe B

1) Earth Sun

$$\begin{aligned} \text{Equatorial radius} &= 5.96 \times 10^8 \text{ m} \\ \text{Mean radius} &= 6.96 \times 10^8 \text{ m} \\ \text{Polar radius} &= 7.96 \times 10^8 \text{ m} \end{aligned}$$

So $R = 5.96 \times 10^8 \text{ m}$

$$\Delta R = 2.0 \times 10^8 \text{ m}$$

and $R \Delta R = 1.39 \times 10^{17} \text{ m}^2 - (8)$

Earth sun distance $= 1.49 \times 10^{11} \text{ m}$

So

$$\begin{aligned} \Delta \theta (\text{slate}) &= 4.73 \times 10^{-5} \text{ radians per orbit} \\ \Delta \theta (\text{EGR}) &= 1.87 \times 10^{-7} \text{ radians per orbit} \end{aligned}$$

—(9)

2) Earth Moon

Using case use :

$$R \Delta R (\text{earth}) = 8.37 \times 10^7 \text{ m}^2 - (10)$$

Earth Moon distance $= 3.844 \times 10^9 \text{ m} - (11)$

So

$$\begin{aligned} \Delta \theta (\text{slate}) &= 4.27 \times 10^{-11} \text{ radians per orbit} \\ \Delta \theta (\text{EGR}) &= 2.20 \times 10^{-11} \text{ radians per orbit.} \end{aligned}$$

—(12)

3) Earth/Gravity Probe B

For note 240(a):

$$\Delta\theta(\text{slate}) = 2.56'' \text{ per orbit}$$

$$\Delta\theta(\text{EGR}) = 2.43 \times 10^{-3}'' \text{ per orbit.} \quad - (13)$$

Use:

$$1 \text{ radian} = 2.06265 \times 10^5'' \quad - (14)$$

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$$\begin{aligned} \Delta\theta(\text{slate}) &= 1.24 \times 10^{-5} \text{ radians per orbit} \\ \Delta\theta(\text{EGR}) &= 1.18 \times 10^{-8} \text{ radians per orbit.} \end{aligned}$$

- (15)

These results show already that a classical calculation would be changed greatly by EGR if the latter were applied logically to all phenomena. The largest effect is the earth moon system, where the classical result would be changed from 4.27×10^{-11} radians per orbit to 6.47×10^{-11} radians per orbit. The orbital period of the moon is 27 days. So in 27 days the moon advances by 4.27×10^{-11} radians due to the slate shape of the earth. In one year of 365 days it would advance by 5.77×10^{-10} radians.

4) It would take 1.09×10^{10} years for the moon to advance by 2π radians. The relativistic correction would change this to 1.65×10^{10} years.

For the earth/sun system the classical precession of the earth due to the oblate nature of the sun is 4.73×10^{-5} radians per orbit, and the EGR effect changes this to $(4.73 + 0.0187) \times 10^{-5} = 4.749 \times 10^{-5}$ radians per orbit. In the classical calculation it would take $2\pi / (4.73 \times 10^{-5}) = 132,837$ years for the earth to precess by 2π radians, but in the relativistic calculation it would take 133,371 years.

To make matters worse Fitzgerald gives a different formula to eq. (1) for the precession of the moon around the earth:

$$V(r) = -\frac{GmM}{r} + \frac{Gm(I_{||} - I_{\perp})P_2 \cos(\gamma_m)}{r^3} \quad (16)$$

where m is the mass of the moon, M is the mass of the earth and γ_m is the angle subtended by the earth's angular velocity vector and I_m , the position of the moon relative to the earth.

5) Eq. (16) gives an orbital precession:

$$\Delta\theta = \frac{6\pi (\underline{I}_\perp - \underline{I}_\parallel) P_2(\cos \delta_m)}{\underline{M} r^3} \quad - (17)$$

where:

$$\underline{I}_\parallel = 8.034 \times 10^{37} \text{ kg m}^2 \quad - (18)$$

$$\underline{I}_\perp = 8.008 \times 10^{37} \text{ kg m}^2 \quad - (19)$$

$$\underline{M} = 6.05 \times 10^{24} \text{ kg} \quad - (20)$$

$$r = 3.844 \times 10^9 \text{ m} \quad - (21)$$

So

$$\Delta\theta(\text{orbital}) = -5.48 \times 10^{-8} \text{ rad per orbit}$$

This is really three orders of magnitude greater than formula (1), although both formulae are derived to describe the same phenomenon. It has been assumed for the sake of simplicity that:

$$P_2(\cos \delta_m) \sim 1 \quad - (22)$$

As in note 240(8), Fitzpatrick gives a formula analogous to Eq. (16) for the precession of the Earth around the sun due to nutation:

$$\phi = -\frac{\underline{m} \underline{M} b}{r} + \frac{\underline{M} b (\underline{I}_\parallel - \underline{I}_\perp)}{2r^3} \left(\frac{3}{2} \sin^2 \theta - 1 \right) \quad - (23)$$

Eq. (23) results in:

b) $\Delta\theta(\text{obs}) = -3.65 \times 10^{-11}$ radians per orbit
 -(24)

which is seven orders of magnitude smaller than
 the result (a).

This means that the rate of change of the sun
 is by far the greater effect. This causes
 the sun to precess by 4.73×10^{-5} radians
 per orbit. This is $9.76''$ per orbit, i.e.
 976 arc seconds per century. The relativistic
 effect is 3.85 arc seconds per century.

Finally the equivalent effects of the planet
 Mercury of the rate of change of the sun are
 determined by the distance of Mercury from the sun, which
 is 0.579×10^{10} m, compared with 1.49×10^{10} m for the earth. So for Mercury:

$\Delta\theta(\text{obs}) =$	6,463	arc seconds per century
$\Delta\theta(\text{EGR}) =$	43	" " "

-(25)

due to the rate of change of the sun

7) Conclusion

In every classical calculation of EGR correction must be applied self consistently. In these examples it has been shown that the EGR correction sometimes has a very large effect. In standard physics it is applied only to the potential:

$$V(\text{EGR}) = -\frac{mMG}{r} - \frac{L^2 m G}{2c^2 r^3} \quad (26)$$

and to no other situation at all. This is obviously irrational and self inconsistent.
